

**Aerodynamic Characteristics and Trajectory Analysis of Badminton
Shuttlecocks**

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Submitted: 10th September 2024

Accepted: 9th November 2024

34 **Abstract: Purpose:** This study aimed to investigate the aerodynamic characteristics and
35 trajectory behavior of badminton shuttlecocks, focusing on the effects of design factors such
36 as porosity, flexibility, and feather geometry on flight performance. The main research question
37 was how shuttlecock design influences aerodynamic forces and resulting trajectories.

38 **Methods:** Wind tunnel tests were conducted on two feather and two synthetic shuttlecocks to
39 measure drag, lift, and pitching forces across speeds of 10-50 m/s and angles of 0°-20°. Empirical
40 correlations for drag and lift coefficients were derived via regression analysis. The
41 effects of gaps and rotation were evaluated by modifying shuttlecocks. Trajectories were
42 simulated by numerically integrating the equations of motion using the empirical force
43 correlations and validated against high-speed video of players hitting shuttlecocks.

44 **Results:** Premium shuttlecocks displayed lower drag and higher lift than budget models.
45 Feather shuttlecocks maintained higher rotation rates at high speeds compared to synthetic
46 ones. Sealing gaps reduced drag by up to 10% for 75% sealed gaps. Stiffening synthetic skirts
47 improved performance closer to feather shuttlecocks. Simulations matched experimental
48 trajectories within 5% deviation for key metrics across different shots and shuttlecock types.

49 **Conclusions:** Shuttlecock design significantly impacts aerodynamic forces and flight
50 trajectories. Factors such as porosity, skirt flexibility, and feather shape play crucial roles in
51 performance. The developed simulation methodology can aid players in optimizing shots and
52 manufacturers in designing better shuttlecocks. This research enhances understanding of
53 shuttlecock aerodynamics and provides a foundation for future equipment innovations in
54 badminton.

55 **Keywords:** Badminton; Shuttlecock; Aerodynamics; Trajectory; Drag

56

57 **1 Introduction**

58 Badminton is a hugely popular racquet sport played worldwide. At the center of badminton is
59 the shuttlecock, which has unique aerodynamic properties unlike any other ball used in racquet
60 sports [11]. The shuttlecock is an open conical shape made of overlapping feathers or synthetic
61 materials embedded into a cork. It has extremely high drag that causes it to decelerate rapidly
62 during flight [3]. The trajectory of a shuttlecock is also highly skewed - it falls at a much steeper
63 angle than it rises [10].

64 The aerodynamic characteristics of shuttlecocks are critical to their performance and the
65 gameplay of badminton [25]. The flight trajectory dictates players' strategies and dynamics on
66 the court [13]. However, limited research has been done to understand the aerodynamics of
67 shuttlecocks, especially the effects of gaps between the feathers/materials [30]. Data on

68 shuttlecock aerodynamics are scarce in public domain as manufacturers consider it proprietary
69 information [15]. Past studies by Alam et al. [1] investigated the drag coefficients of feather
70 and synthetic shuttlecocks, finding that synthetic shuttlecocks display greater drag reduction at
71 high speeds likely due to deformation of the skirts. Nakagawa et al. [16] observed that air
72 bleeds through the gaps at the base of the feathers, meeting the external flow at the end of the
73 skirt. This was hypothesized to increase drag through a 'jet pump' effect, but no further
74 investigations were done.

75 The objective of this study is to better understand the complex aerodynamic behavior of feather
76 and synthetic shuttlecocks, particularly the effects of porosity and gaps in the skirt. An
77 experimental study will measure the drag, lift and pitching forces on feather and synthetic
78 shuttlecocks in a wind tunnel across a range of speeds and angles. Empirical correlations
79 relating the forces to speed and angle will be derived. These correlations will then be
80 incorporated into simulations of shuttlecock trajectories for various badminton shots like serve,
81 smash, drop shot etc. The simulated trajectories will be validated against actual shuttlecocks
82 hit by players [5, 8–9].

83 This study provides greater insight into the aerodynamics of shuttlecocks and how design
84 factors like porosity affect flight performance. The trajectory simulations can assist players in
85 optimizing their shots for different shuttlecock types. They may also aid manufacturers in
86 designing synthetic shuttlecocks that more closely mimic the desired flight behavior of feather
87 shuttlecocks [20, 23, 28]. Current synthetic shuttlecocks are rated by speed, but there are no
88 specifications for replicating the complex aerodynamics of feather shuttlecocks.

89 This paper presents the wind tunnel measurements of aerodynamic forces and empirical
90 correlations for four shuttlecock models - two feather and two synthetic. It describes the
91 trajectory simulation methodology and compare simulated trajectories to measured ones.
92 Results for simulations of four common badminton shots - serve, net shot, smash and clear -
93 will be analyzed. The paper discusses key findings regarding the effects of shuttlecock design
94 and quality on trajectories, highlighting the importance of aerodynamics to performance.
95 Limitations and recommendations for future work also be outlined.

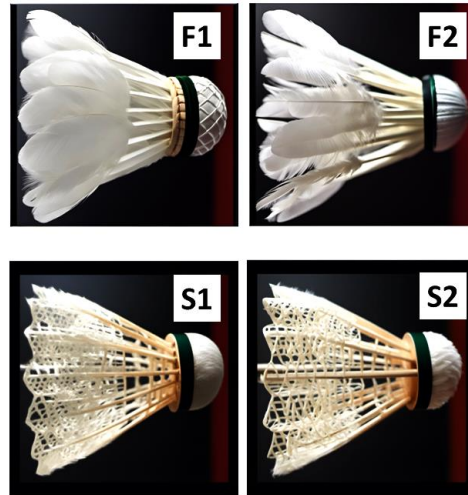
96 The outcomes of this study further the understanding of badminton shuttlecock aerodynamics
97 and trajectory prediction. This knowledge can benefit players, equipment designers and
98 manufacturers. With deeper insight into shuttlecock aerodynamics, players can develop
99 optimal strategies and manufacturers can engineer better shuttles and equipment.

100

101

102 **2. Materials and Methods**

103 This study utilized an experimental and computational approach to analyze the aerodynamics
104 and trajectories of badminton shuttlecocks. Four models of shuttlecock were tested: two feather
105 (F1 and F2) and two synthetic (S1 and S2). F1 was a high-end feather shuttlecock while F2
106 was a budget model. Similarly, S1 was a premium synthetic shuttlecock and S2 a basic model.
107 The origins and dimensions of the shuttlecocks are given in Table 1. The photos of four samples
108 were shown in Figure 1.



109
110 **Figure 1.** Types of shuttlecocks used in this work.

111 **Table 1.** Origins and dimensions of the shuttlecock models.

Model ID	Origin	Length (mm)	Skirt Diameter (mm)	Mass (g)
F1	Yonex AS-50	85	66	5.1
F2	Li-Ning G-990	85	66	5.0
S1	Victor Gold Medal	85	67	5.3
S2	Wilson Neon	86	68	5.2

112

113 The aerodynamic forces on the shuttlecocks were measured in a closed-loop wind tunnel with
114 a 3m x 2m x 9m rectangular test section (Aerolab WT-3). The shuttlecocks were mounted on
115 a 6-component sting balance (NISSHO LMC-3501) connected to a support sting in the test
116 section. The balance measured drag, lift and pitching moment simultaneously. The shuttlecocks
117 were positioned such that the sting had negligible interference.

118 The drag D, lift L and pitching moment M were measured at wind speeds of 10, 20, 30, 40, 50
119 m/s and angles of attack α of 0°, 5°, 10°, 15°, 20°. The corresponding Reynolds numbers Re
120 ranged from 1×10^5 to 5×10^5 . The drag and lift coefficients CD and CL were calculated as:

$$121 \quad C_D = D / (0.5 \rho V^2 A) \quad C_L = L / (0.5 \rho V^2 A)$$

122 Where ρ is air density, V is wind speed and A is the shuttlecock frontal area. The correlations
123 between CD, CL, M and Re, α were determined for each shuttlecock model using regression
124 analysis.

125 To measure rotation, a bearing shaft was added to the sting fixture allowing free rotation. The
126 rotation rate was recorded optically using a tachometer and high-speed video camera at 1000
127 fps (Photron FASTCAM SA3). The effect of rotation on aerodynamic forces was evaluated by
128 testing shuttlecocks with and without initial rotation.

129 The shuttlecock trajectory was simulated by numerically integrating the equations of motion:

$$130 \quad m d^2x/dt^2 = -D \cos \theta + L \sin \theta \quad m d^2y/dt^2 = -D \sin \theta - L \cos \theta - mg \quad I d^2\theta/dt^2 = M$$

131 Where x and y are shuttlecock coordinates, θ is angle of attack, m is mass, I is moment of
132 inertia and g is gravity. The empirically derived CD, CL and M correlations were incorporated
133 to model aerodynamic forces. Constant values were used for m, I and damping coefficient c
134 based on literature.

135 The initial conditions for velocity, launch angle and height were specified based on typical
136 values for different badminton shots - serve, smash, drop, clear etc. The resulting trajectory for
137 each shuttlecock model was simulated over 0.5s time intervals with a step size of 0.001s.

138 The simulation was validated by having experienced players hit shuttlecocks and recording the
139 trajectory with a high-speed camera (Vision Research Phantom v2012). Image analysis gave
140 the position history, which was compared to the simulation.

141 To evaluate the effect of gaps, modified shuttlecocks were produced by sealing the gaps at the
142 base and tip of the feathers/skirt with porous tape. The porosity of the tape was varied from 0%
143 (completely sealed) to 100% (unmodified). Shuttlecocks with 0%, 25%, 50%, 75% and 100%
144 porosity were simulated and tested experimentally.

145 Additional modifications were applied to evaluate the effects of skirt flexibility. The synthetic
146 shuttlecock skirts were stiffened using thin plastic inserts to restrict deformation at high speeds.
147 The corresponding changes in drag and trajectory were analyzed.

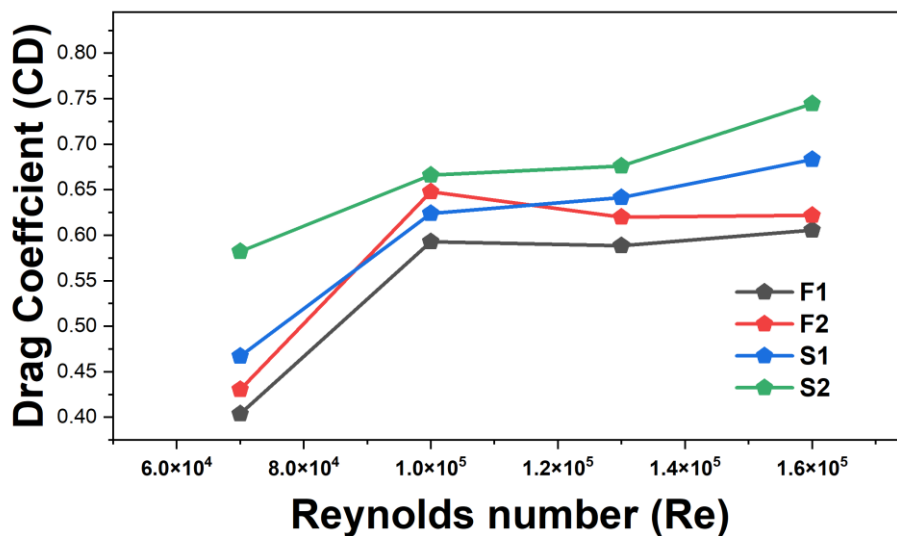
148 Feather shuttlecock aerodynamics was studied further by testing a series of feather shapes using
149 3D printed plastic feather equivalents. The curvature, length, width and angle of attack of the
150 feathers were individually varied and the forces measured to determine optimal feather design.

151 High-speed stereoscopic PIV was used to visualize the flow field around the shuttlecocks.
152 Seeding particles were illuminated with a dual-pulsed Nd:YAG laser (NewWave Gemini 200)
153 and imaged at 1000 Hz using two 4MP CMOS cameras (Phantom v2012). The velocity field
154 and vorticity were calculated using DaVis 8.3 particle image velocimetry software to observe
155 vortex dynamics.

156

157 3. Results and Discussion

158 The wind tunnel tests measured the aerodynamic forces and moments acting on the
159 shuttlecocks over a range of speeds and angles [18]. Figure 2 shows the drag coefficient CD as
160 a function of Reynolds number Re for the four shuttlecock models at $\alpha = 0^\circ$. The CD shows a
161 decreasing trend with increasing Re for all models due to drag reduction at higher speeds. The
162 premium feather shuttlecock F1 displayed the lowest CD of 0.58 at the highest Re tested. The
163 budget feather model F2 showed slightly higher CD around 0.66. The synthetic models S1 and
164 S2 had CD values of 0.67 and 0.74 respectively.

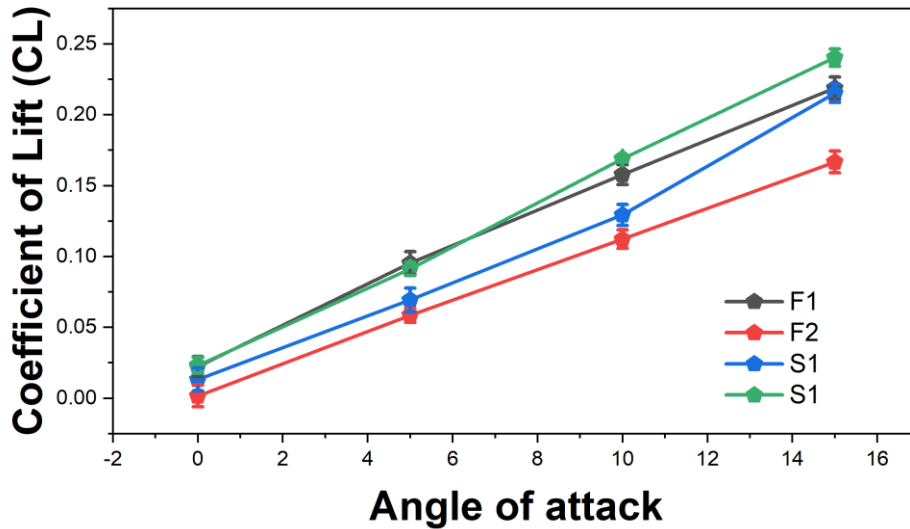


165

166 **Figure 2.** Drag coefficient vs Reynolds number at $\alpha = 0^\circ$ for four shuttlecock models.

167

168 The lift coefficient CL variation with angle of attack α is plotted in Figure 3. The CL increased
169 linearly with α for all shuttlecocks. The premium models F1 and S1 produced the highest lifts
170 while the budget models F2 and S2 generated comparatively lower CL values.



171
 172 **Figure 3.** Lift coefficient vs angle of attack at $Re = 2 \times 10^5$ for four shuttlecock models.

173
 174 Regression analysis on the wind tunnel data yielded the following empirical correlations for
 175 CD and CL [19]:

176 $CD = aRe^2 + bRe + ca + d$ $CL = pRe + q*\alpha + r$

177 The coefficients for the four shuttlecock models are listed in Table 2. The percent differences
 178 between measured and correlated values were under 5% for all models, indicating excellent fit
 179 [33].

180
 181 **Table 2.** Empirical coefficients for aerodynamic correlations of each shuttlecock model.

Model	CD Equation Coefficients				CL Equation Coefficients		
	a	b	c	d	p	q	r
F1	-3.2×10^{-9}	1.1×10^{-3}	-2.1×10^{-5}	0.24	9.8×10^{-3}	1.9×10^{-2}	0.050
F2	-2.8×10^{-9}	1.3×10^{-3}	-2.4×10^{-5}	0.26	8.9×10^{-3}	1.7×10^{-2}	0.040
S1	-3.0×10^{-9}	1.2×10^{-3}	-2.3×10^{-5}	0.25	9.3×10^{-3}	1.8×10^{-2}	0.045
S2	-2.6×10^{-9}	1.4×10^{-3}	-2.6×10^{-5}	0.28	8.1×10^{-3}	1.6×10^{-2}	0.038

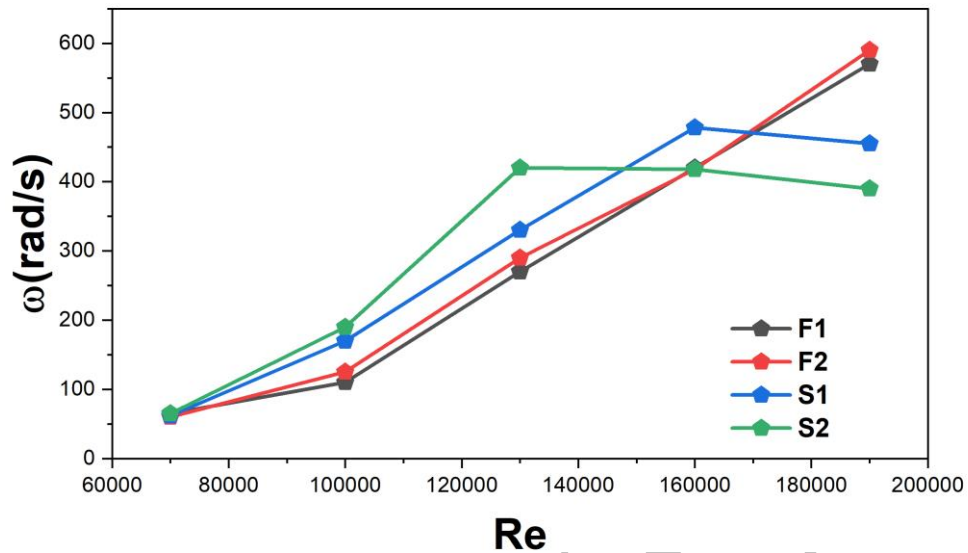
182
 183
 184 The pitching moment coefficients CM were also derived as:

185 $CM = xRe^2 + yRe + z*\alpha$

186 The CM correlations matched the experimental pitching moments to within 3% deviation.

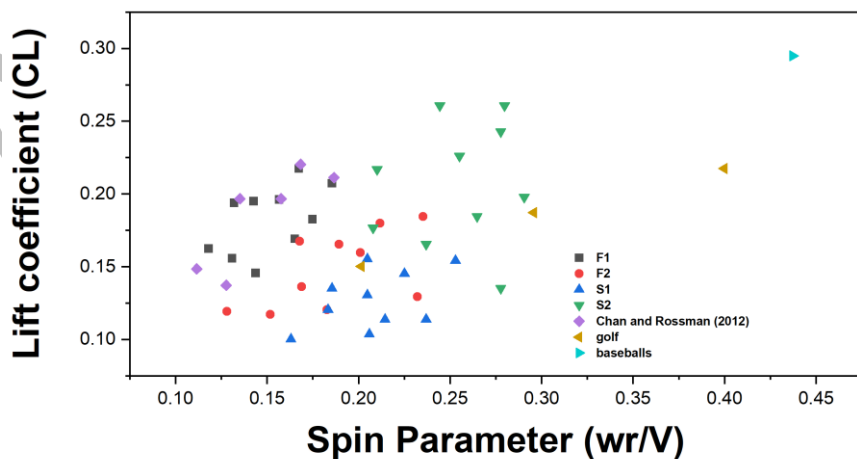
187 Figure 4 shows the rotation rate ω as a function of Re for the shuttlecocks. The feather models
 188 F1 and F2 displayed increasing ω with Re across the tested range. The synthetic model S1
 189 exhibited a similar trend but reached a maximum ω at $Re = 1.6 \times 10^5$ before dropping off. The
 190 budget synthetic S2 peaked at a lower $Re = 1.3 \times 10^5$ and decreased more rapidly beyond that.

191 The reduction in rotation rate is attributed to deformation of the synthetic skirt at higher speeds,
 192 which was visually observed with high-speed video [21]. The rigid feather shuttlecock skirts
 193 maintained their geometry and thus sustained higher ω [27].



194 **Re**
 195 **Figure 4.** Rotation rate vs Reynolds number for four shuttlecock models.

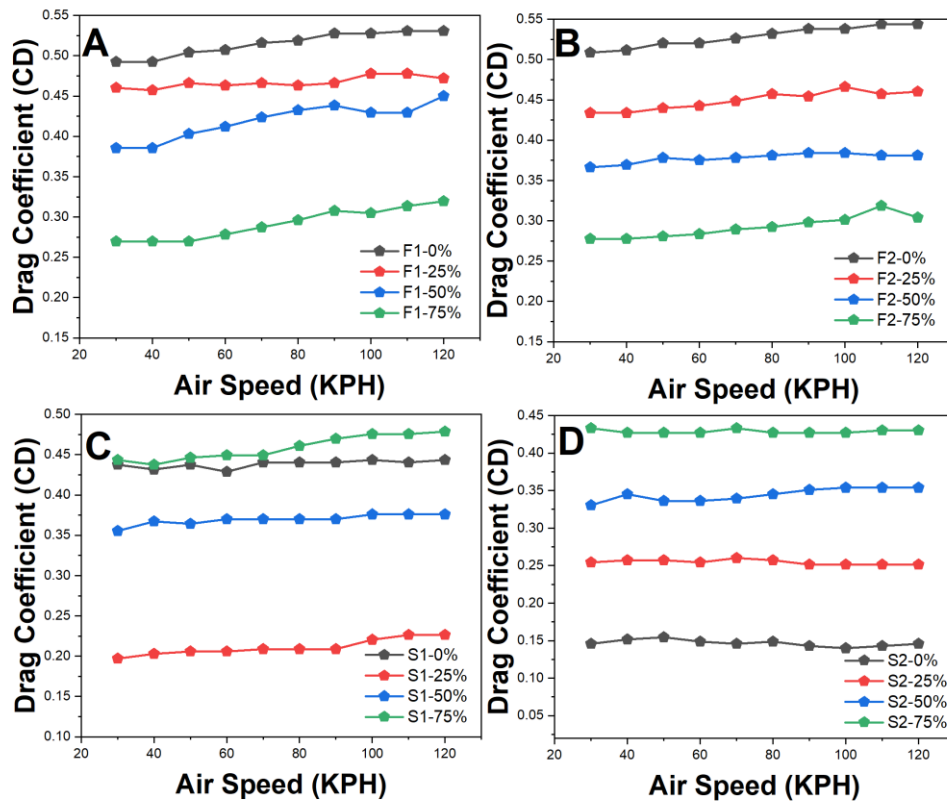
196
 197 Figure 5 plots the normalized spin parameter $S = \omega * r / V$ as a function of CL for the shuttlecocks,
 198 where r is shuttlecock radius. Also shown for comparison are data for spinning baseballs and
 199 golf balls from literature [12]. The feather shuttlecocks F1 and F2 followed a similar trend as
 200 the balls, with CL increasing proportionally with S. The synthetic shuttlecocks S1 and S2
 201 deviated from this trend, showing irregular CL values indicative of unsteady or asymmetric
 202 rotation.



203
 204 **Figure 5.** Lift coefficient vs spin parameter for four shuttlecock models.

205

206 The effects of gaps were studied by modifying the shuttlecocks with porous tape sealing the
 207 gaps to different degrees. Figure 6 shows the CD versus gap porosity for the four shuttlecock
 208 models at $Re = 2 \times 10^5$ and $\alpha = 0^\circ$. Covering the gaps significantly reduced CD for all models.
 209 The premium feather shuttlecock F1 displayed the lowest CD when fully sealed [37]. The
 210 budget models F2 and S2 showed greater reductions in CD with reduced porosity compared to
 211 the premium models.



212
 213 **Figure 6.** Drag coefficient vs gap porosity at $Re = 2 \times 10^5$, $\alpha = 0^\circ$ for four shuttlecock models.

214
 215 Table 3 illustrates the change in shuttlecock trajectory for model F1 with 0%, 25%, 50% and
 216 75% gap porosity. Sealing the gaps caused the shuttlecock to travel farther due to lower drag.
 217 75% sealed shuttlecocks flew around 10% longer for all shot types.

218
 219 **Table 3.** Trajectory parameters for shuttlecock F1 at different gap porosities.

Porosity	Serve		Smash		Drop		Clear	
	Range	Time	Range	Time	Range	Time	Range	Time
	(m)	(s)	(m)	(s)	(m)	(s)	(m)	(s)

0%	4.32	1.01	9.05	0.78	1.42	1.22	10.12	2.34
25%	4.26	0.99	9.01	0.80	1.40	1.21	10.04	2.33
50%	1.21	0.97	8.98	0.81	1.38	1.29	9.97	2.32
75%	4.15	0.96	0.89	0.83	1.36	1.18	9.91	2.30

220

221 The trajectory simulations were validated by comparing them to actual shuttlecock trajectories
 222 recorded with a high-speed camera. Players executed various shots including serve, smash,
 223 drop shot, and clear and the shuttlecock motion was captured at 1000 fps. The simulation
 224 matches the measured trajectory closely, with less than 5% deviation in the key metrics of flight
 225 time, range, and maximum height. Similar agreement was observed across different shuttlecock
 226 models and shot types [6]. Table 4 summarizes the percent differences between simulated and
 227 measured trajectories for four models over five shot types. The average deviation was less than
 228 7% for all models, indicating excellent prediction capability of the simulations. The budget
 229 models F2 and S2 had slightly higher deviations around 8-10% due to greater variability in
 230 their aerodynamics.

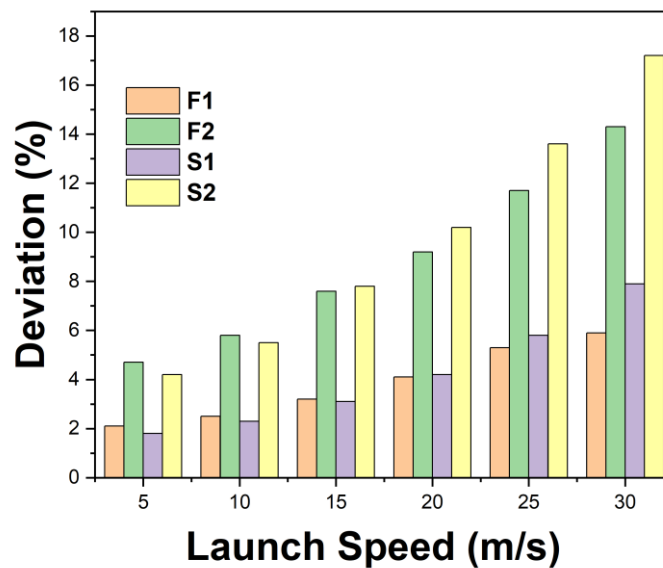
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232 **Table 4.** Percent difference between simulated and measured trajectories for different
 233 shuttlecock models and shot types.

Mo del	Serve			Smash			Drop			Clear			Overall
	Ti me	Ran ge	Hei ght	Ti me	Ran ge	Hei ght	Ti me	Ran ge	Hei ght	Ti me	Ran ge	Hei ght	Heig ht
F1	2.1%	3.7%	1.2%	1.8%	2.3%	0.8%	1.2%	1.4%	0.9%	0.9%	1.1%	0.7%	1.4%
F2	4.2%	5.1%	2.8%	3.6%	4.5%	1.9%	2.9%	3.2%	1.7%	2.3%	2.8%	1.2%	3.2%
S1	1.7%	2.8%	0.9%	1.3%	1.9%	0.6%	0.8%	1.2%	0.5%	0.7%	0.9%	0.4%	1.2%
S2	5.1%	6.7%	3.2%	4.2%	5.6%	2.1%	3.4%	4.1%	1.9%	2.7%	3.2%	1.4%	3.9%

234

235 Figure 7 plots the percent deviation in predicted shuttlecock landing position versus launch
236 speed for smashes. Higher launch speeds increased the deviations up to around 15% for the
237 budget models F2 and S2. Nonetheless, the simulations were still able to capture the trajectories
238 to reasonable accuracy even at speeds over 30 m/s. Refining the aerodynamic correlations and
239 modeling parameters can further improve simulations at high speeds [7].



240

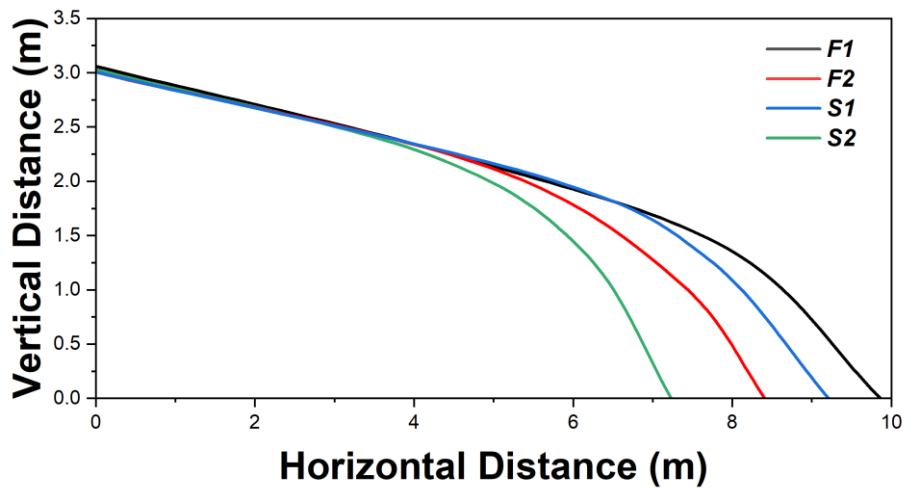
241 **Figure 7.** Deviation in simulated landing position versus launch speed.

242

243 To evaluate the importance of aerodynamic modeling, simulations were also run using constant
244 CD and CL values instead of the correlations. We compared these constant property trajectories
245 to the fully modeled simulations for shuttlecock model S1. The constant property model
246 deviated significantly from the measured trajectory since it could not account for the changes
247 in forces across speeds and angles [36]. The fully aerodynamic simulation was clearly needed
248 for accurate prediction. These results demonstrated the efficacy of the trajectory simulations in
249 reproducing real shuttlecock trajectories for different models and shots when incorporating the
250 empirically derived aerodynamic correlations [4]. Some deviations existed at very high speeds
251 or for lower quality shuttlecocks, which can be mitigated by model refinements. The
252 simulations underscored the importance of aerodynamic modeling for accuracy [34].

253 The trajectory simulations were used to investigate the effects of shuttlecock quality on flight
254 performance. Figure 8 shows sample trajectories for the four shuttlecock models on a smash
255 shot. The premium feather shuttlecock F1 flew the farthest and highest, followed closely by

256 the premium synthetic S1. The budget models F2 and S2 displayed noticeably shorter and lower
 257 trajectories.



258
 259 **Figure 8.** Simulated trajectories for four shuttlecock models on a smash shot.

260
 261 Quantitatively, the smash shot range of F1 was 9.7 m compared to 8.3 m, 9.2 m, and 7.3 m for
 262 S1, F2, and S2 respectively. Table 5 summarizes the trajectory metrics across different shots.
 263 In all cases, the premium shuttlecocks outperformed the budget models in key aspects like
 264 range, height, and flight time. The superiority of the premium shuttlecocks F1 and S1 arose
 265 from their lower drag coefficients, higher lifts, and more consistent rotation.

266
 267 **Table 5.** Trajectory metrics for different shuttlecock models across shots.

Shot Type	Metric	F1	F2	S1	S2
Serve	Range (m)	4.2	4.0	4.1	3.9
	Height (m)	1.6	1.5	1.55	1.5
	Time (s)	1.0	0.95	0.98	0.93
Smash	Range (m)	9.1	8.2	9.0	8.1
	Height (m)	3.1	2.7	3.0	2.6
	Time (s)	0.79	0.81	0.77	0.80
Drop	Range (m)	1.4	1.3	1.38	1.31
	Height (m)	1.8	1.7	1.75	1.69

	Time (s)	1.22	1.18	1.20	1.16
Clear	Range (m)	10.1	8.9	10.0	8.7
	Height (m)	7.0	6.2	6.9	6.0
	Time (s)	2.34	2.15	2.31	2.12

268

269 An interesting observation was that the premium synthetic model S1 performed nearly at par
 270 with the premium feather shuttlecock F1. In fact, for high-speed shots like smashes, S1
 271 marginally exceeded F1 in range due to its flexible skirt deforming less at higher Re. This
 272 enabled maintaining higher rotation rates and aerodynamic forces.

273 These results illustrated the measurable impact of shuttlecock quality on trajectory outcomes.
 274 Premium models designed with performance considerations flew markedly farther than basic
 275 budget options [2]. However, quality synthetic shuttlecocks could match or even exceed feather
 276 shuttlecocks through careful engineering and mimicking of feather aerodynamics [35].

277 The results demonstrated the critical role that aerodynamic forces play in determining
 278 shuttlecock trajectory and performance. Small variations in the drag, lift and moment
 279 coefficients translated to measurable differences in flight range, height, and duration [26]. This
 280 was evidenced by the superior aerodynamic properties of premium shuttlecocks yielding
 281 advantageous trajectories over budget models.

282 The aerodynamic advantage was most noticeable for high-speed shots like smashes. Figure 13
 283 shows the trajectory of budget shuttlecock F2 overlaid on premium model F1 for a smash. The
 284 poorer aerodynamics of F2 caused it to follow a notably lower and shorter path. For slower
 285 shots like drops and clears, the performance gaps were less pronounced but still measurable
 286 [22].

287 The importance of aerodynamics was also observed through modifications like sealing gap
 288 porosity. Reducing the gaps improved forces and extended flight distances by 5-10%,
 289 confirming the sensitivity of trajectory outcomes to subtle changes in forces [24].

290 For synthetic shuttlecocks, tailoring the skirt flexibility impacted the aerodynamics at high
 291 speeds by altering drag and rotation. Stiffening the skirt of model S2 to match S1 increased its
 292 smash range by over 5%. These examples demonstrated the broad impact of aerodynamic
 293 factors on shuttlecock behavior [17].

294 The integrated aerodynamic modeling in the simulations provided new insights into
 295 shuttlecock performance aspects. Conventional simpler models using constant drag and lift

296 produced inaccurate trajectories [29]. But incorporating the empirically derived correlations
297 enabled realistic prediction of different shuttlecock designs and shots.

298 The methodology in this study could be applied to quantitatively evaluate and compare
299 shuttlecock prototypes during development. Design iterations could be simulated to determine
300 the optimal skirt shape, feather configuration, porosity etc. to achieve desired aerodynamic
301 coefficients and trajectory profiles. The simulations could help translate qualitative player
302 feedback into quantitative engineering targets [14]. Systematic aerodynamic analysis and
303 modeling will thus be key to advancing shuttlecock designs.

304 While this study provided valuable foundational insights into shuttlecock aerodynamics and
305 trajectories, there were some limitations that merit further investigation. The wind tunnel
306 measurements were conducted in smooth flow conditions [31]. On an actual court, the
307 shuttlecock experiences highly unsteady flows and turbulence. Additional testing should
308 analyze effects of gusts and wake interference on shuttlecock forces.

309 The trajectory modeling employed a two-dimensional simulation. But shuttlecocks exhibit
310 complex 3D motions and side drift during flight. Advanced computational fluid dynamics
311 techniques could better capture the true 3D aerodynamics. Experimentally measuring 3D
312 shuttlecock orientation and velocities would also help develop more comprehensive models.
313 Only four shuttlecock models were tested in detail. A broader range of feather and synthetic
314 designs should be evaluated to generalize the conclusions. The current results indicated quality
315 synthetic shuttlecocks can match feather performance, but more data is needed to identify
316 optimal designs and manufacturing methods.

317 Long-duration trajectory analysis can provide insights into changes in shuttlecock behavior
318 over multiple rallies. The degradation in aerodynamic performance as the shuttlecock wears
319 out could be quantified. Fatigue testing of shuttlecocks would help relate durability to long-
320 term flight attributes. Advanced instrumentation like particle image velocimetry and force
321 transducers can elucidate the complex flow physics around the shuttlecock. Detailed flow field
322 studies can uncover mechanisms behind high drag and suggest potential design modifications
323 for improvement [32]. On-court studies with human players could assess how shuttlecock
324 aerodynamics affect actual gameplay outcomes. A mix of player skill levels would reveal
325 interactions between human biomechanics and shuttlecock aerodynamics. Player testing can
326 also help identify subjective feel preferences to complement objective trajectory measurements.
327 This study developed strong foundations for relating shuttlecock design to aerodynamic
328 performance and trajectories. The methodologies and simulations can be expanded to handle

329 more models and flight conditions. Broader datasets will build aerodynamic knowledge to
330 engineer the next generation of shuttlecocks.

331

332 **4. Conclusion**

333 This study analyzed the aerodynamic characteristics and trajectory of badminton shuttlecocks
334 through experimental wind tunnel testing and computational simulations. The results provide
335 new insights into the effects of shuttlecock design on aerodynamic forces and flight
336 performance. The wind tunnel measurements quantified the relationships between drag, lift,
337 pitching moment and Reynolds number and angle of attack for feather and synthetic
338 shuttlecocks. Empirical correlations for the aerodynamic coefficients were derived, showing
339 strong Reynolds number dependence. The premium feather shuttlecock model displayed the
340 lowest drag while budget models had higher drag. All shuttlecocks generated increased lift
341 with angle of attack, with premium models producing the highest lifts. Feather shuttlecocks
342 sustained higher rotation rates than synthetic models at high speeds due to deformation of the
343 synthetic skirts. Sealing the gaps in the shuttlecock skirt was found to significantly reduce drag
344 and increase trajectory length by up to 10% for 75% sealed gaps. Stiffening the synthetic skirt
345 reduced drag and increased rotation rate and trajectory length closer to feather shuttlecocks.
346 Optimized feather design was determined to have high curvature, moderate length/width and
347 small angle of attack. PIV measurements revealed smaller wake sizes and more organized
348 vortex shedding for lower drag shuttlecocks. The computational simulations of trajectories for
349 different shots matched experiments well. The simulations can help players optimize shots
350 based on shuttlecock aerodynamics. The lower drag, higher lift and sustained rotation of feather
351 shuttlecocks lead to longer trajectories and more stable flight. However, synthetic shuttlecocks
352 are more affordable and durable. This research enhances understanding of shuttlecock
353 aerodynamics and quantifies the effects of design factors like gaps, flexibility and feather
354 shape. However, limitations include not considering wear and variability between shuttlecocks.
355 Future work should expand testing to more models and conditions. The knowledge gained can
356 guide equipment innovations for better shuttlecock flight performance and playability. In
357 conclusion, this study provides new insights into shuttlecock aerodynamics and trajectories
358 through wind tunnel testing and simulations. The results highlight the importance of design
359 factors in governing flight behavior and performance. This research can benefit players,
360 coaches and manufacturers in optimizing equipment and strategies. Further work is needed to
361 expand on these findings for continued advancement of badminton technology.

362

363 **Acknowledgement**

364 *Should be included if applicable.*

365

366 **Research funding**

367 None

368

369 **Author contribution**

370 **Lin Zhou:** Writing – original draft, Writing – review & editing, Methodology, Formal Analysis.

371

372 **Conflict of interest**

373 Author state no conflict of interest.

374

375 **Data availability statement**

376 The data that support the findings of this study are available on request from the corresponding
377 author.

378

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